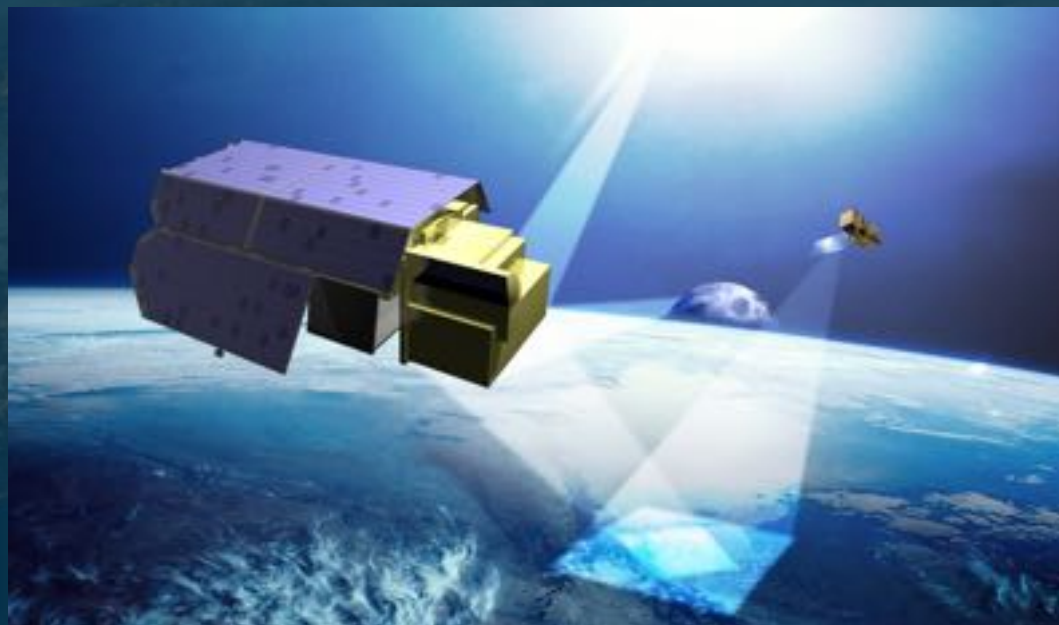




Traceable Radiometry Underpinning Terrestrial- & Helio- Studies

An ESA EarthWatch mission



A 'gold standard' reference in space to support the climate emergency

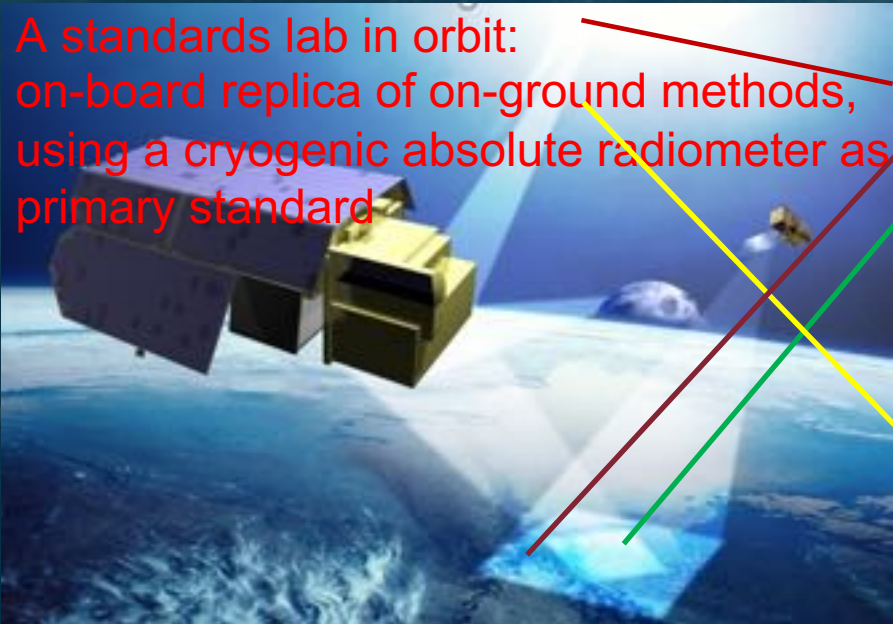
Nigel Fox, Paul Green, Samuel Hunt, NPL,
Andrea Marini, Thorsten Fehr, ESA
Kyle Palmer, Airbus (now ESA)
John Remedios NCEO Univ Leicester
+++++

SITSats and TRUTHS Mission Objectives

*What is a SITSat?: 'Space borne missions specifically designed, characterised and documented to provide **high accuracy SI-Traceable** 'reference' measurements.'* (Evidencing comprehensive uncertainty to SI, 'in-space', of all contributors to observations made from the satellite)

TRUTHS is an **operational climate mission**, aiming to:

A standards lab in orbit:
on-board replica of on-ground methods,
using a cryogenic absolute radiometer as
primary standard



1. **Climate benchmarking:** enhance our ability to estimate the **Earth Radiation Budget** (and attributions) through direct measurements of incoming & outgoing energy and reference calibration of other ERB & similar missions.

2. **Satellite cross-calibration:** establish a 'standards laboratory in space' to create a '**gold standard**' reference data set to cross-calibrate other sensors and improve the quality and interoperability of their data through: simultaneous observations, surface reference sites and the moon

3. provide SI-traceable measurements of the **solar spectrum (incoming & reflected)** to address its impact on climate and interactions with the atmosphere and surface

A **benchmark measurement** is one with characteristics (documentation, SI-Traceable uncertainty, representative sampling) that allows it to be unequivocally considered a 'reference' of the specified measurand against which future measurements of the same measurand, can be compared.

What does TRUTHS do?

Measures incoming & Earth/Moon reflected radiation from the Sun

- 320 to 2400 nm @ ~4 nm intervals (1 nm for solar UV)
- Global nadir @ 50 m ground resolution (capability) with 100 km swath
- Target uncertainty of 0.3% (k=2)

Establishing a benchmark of the radiation state of the planet at ToA (radiance/reflectance) & BoA surf reflectance to help enable:

Observations

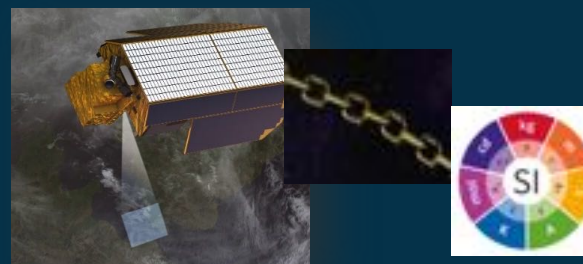
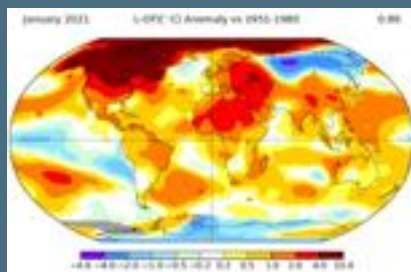
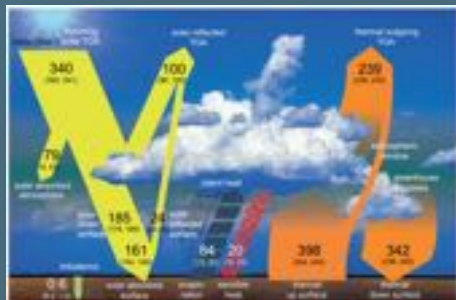
- Benchmark
- monitoring
- Litigation
- algorithm improvement

Calibration

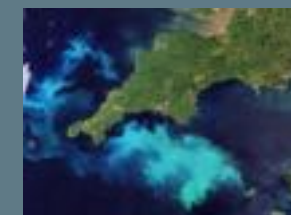
- Interoperability
- data-gaps
- performance
- Utility

Climate action: Supporting 'Net Zero'

Climate sensitivity/response



Climate action/mitigation

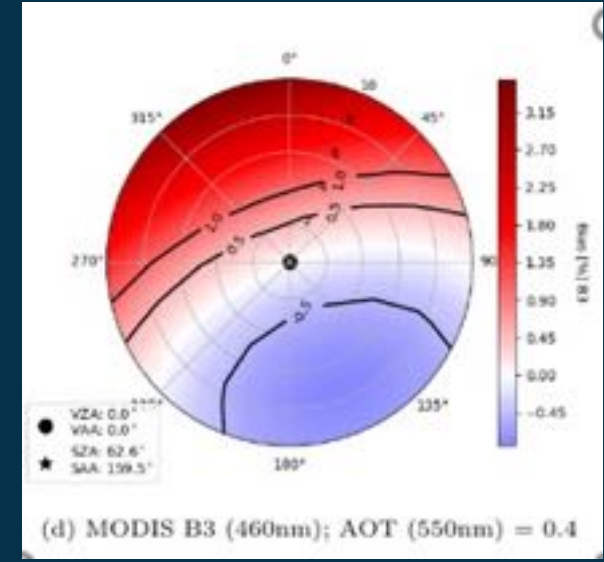
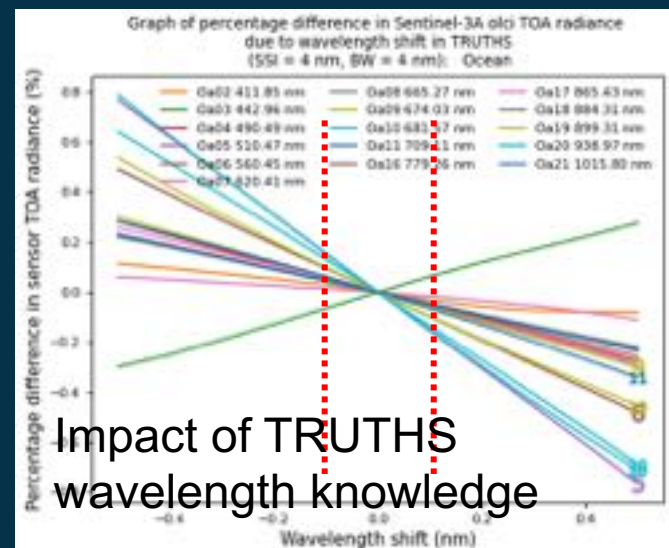
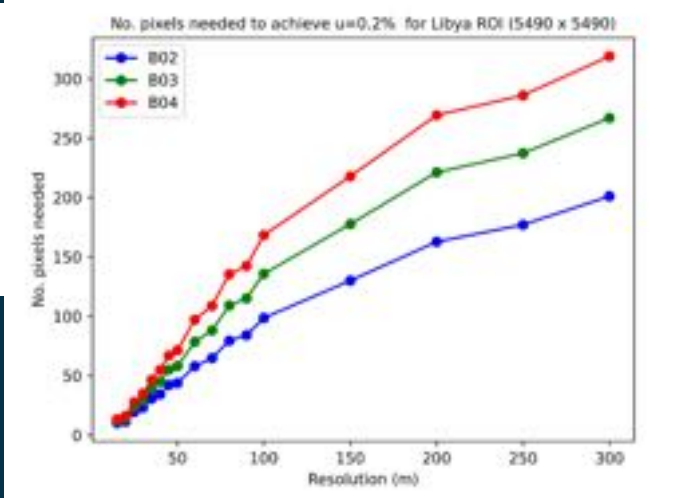
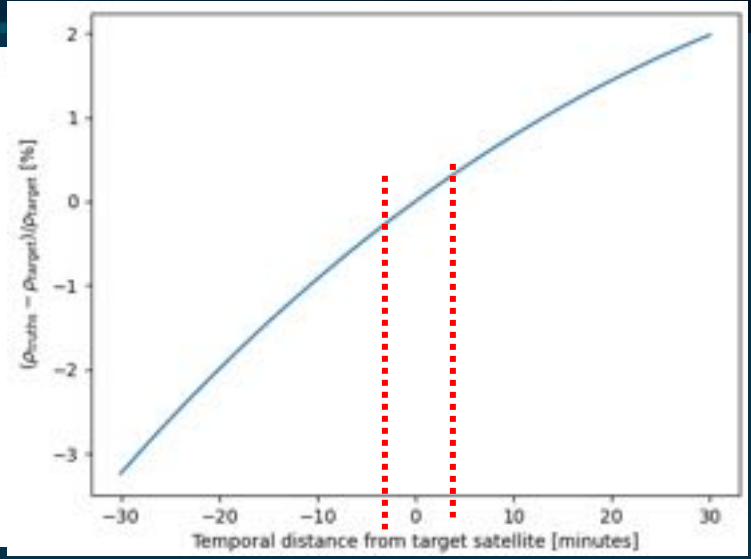
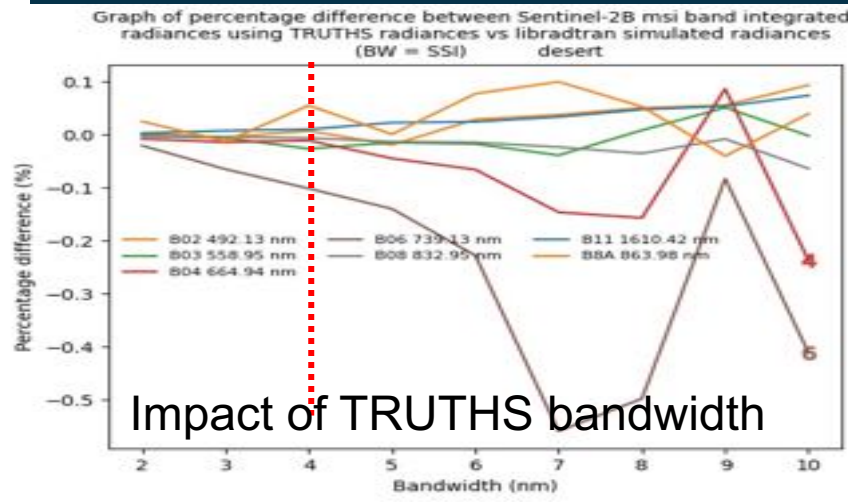
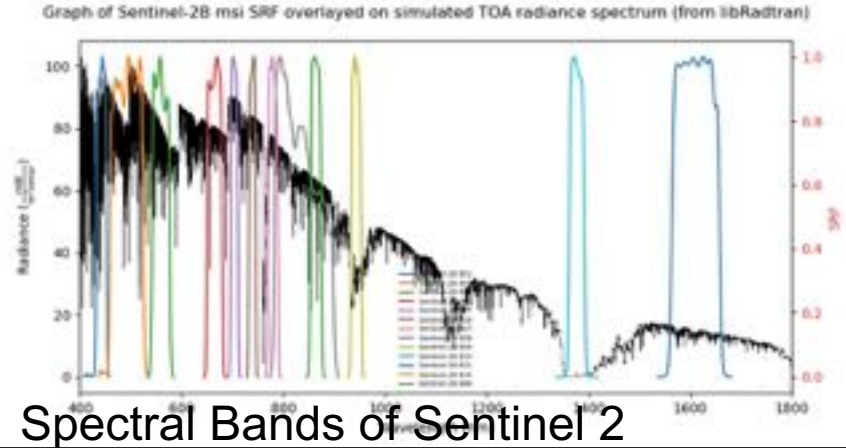


Adaptation/sustainability



Transferring TRUTHS accuracy to other Sensors: establishing mission requirements (S2S calibration)

(Fahy, Hunt, Stedman, Gorrono)



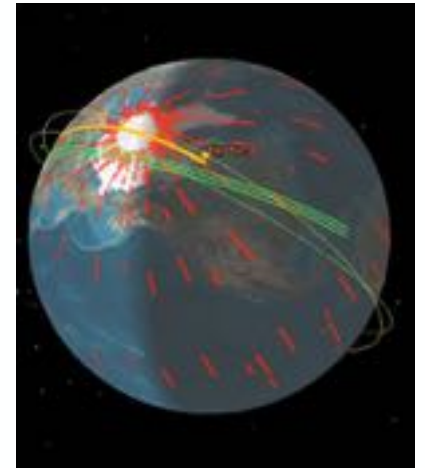
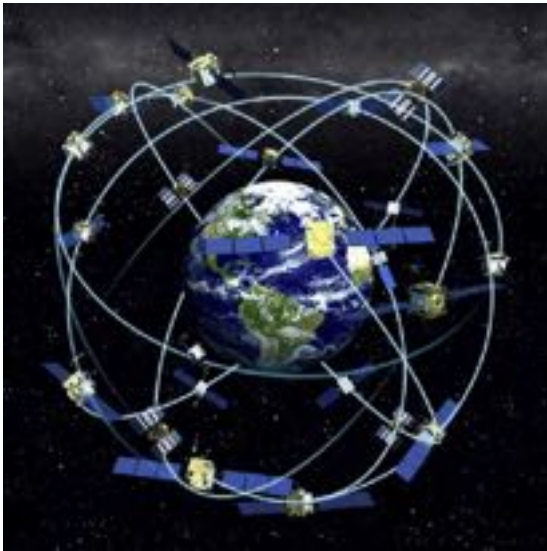
Uniformity of Cal Target (Libya 4) area to be sampled (2.5 km to achieve 0.2% @ 50 m)



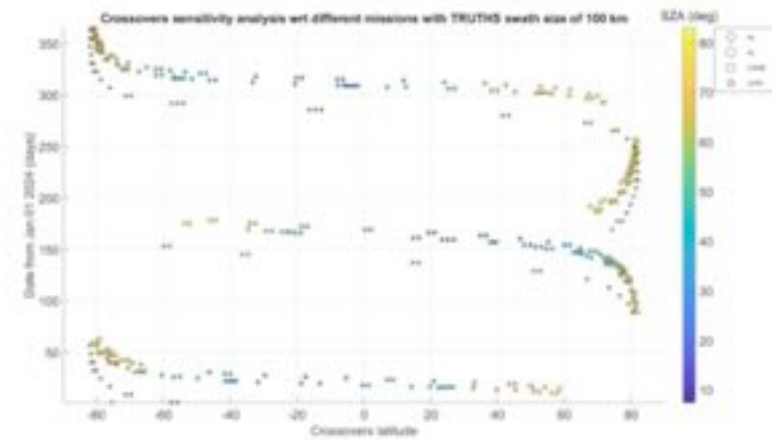
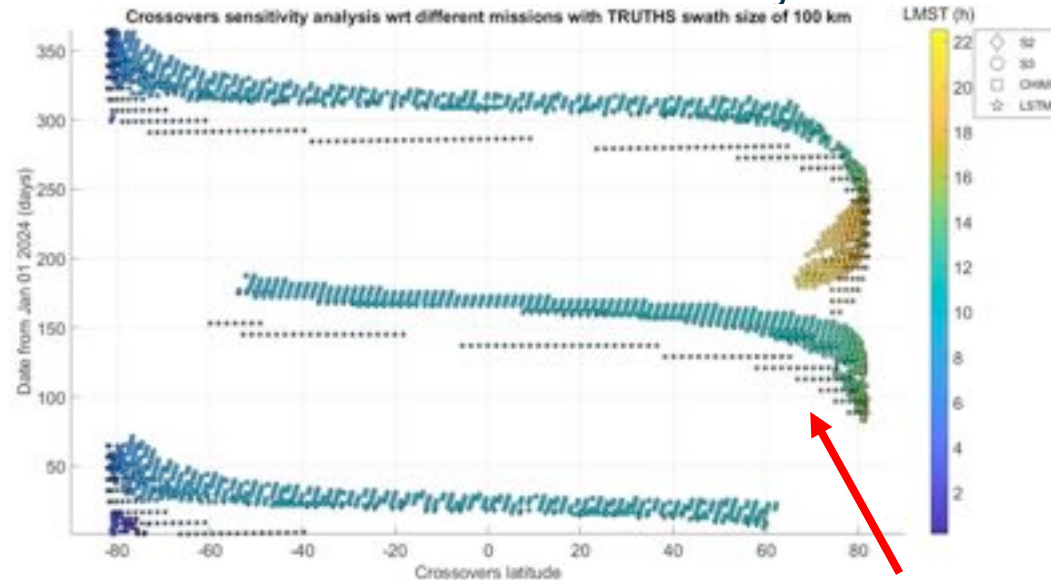
Reference Calibration

- Enables interoperability & Harmonisation
 - Prospect of 'certified calibration'

TRUTHS 90° pole to pole orbit, observing through the diurnal cycle, allows many opportunities to overpass orbit of sun-synchronous sensors



Summary after 6 months



<30 s time difference Swath overlap

1 year of near perfect nadir overlaps for TRUTHS & satellite under test

(<1° (no pointing) <30 s time difference

TRUTHS provides the means to transform global EO system, including constellations of micro-sats so they deliver traceable scientific/climate quality observations -

Uncertainty budget for TRUTHS – satellite comparisons (convolution & Uc demonstrated also for hyperspectral mission)

Uncertainty	Best S2 bands	Worst S2 bands
Spectral resolution TRUTHS	0.1 %	0.6 %
Spectral accuracy TRUTHS	0.1 %	0.2 %
Spatial co-alignment mismatch	0.1 % (Libya) 0.12 % (La Crau)	0.1 % (Libya) 0.5 % (La Crau)
30 minute time difference (atmospheric effects)	0.1 % (if corrected) 0.3 % (if atmosphere not known)	0.1 % (if corrected) 2 % (if atmosphere not known)
30 minute time difference (surface BRF)	0.2 %	0.4 %
Combined with reasonable corrections	0.4 % - 0.5 %	0.7 %

Hyper-spectral applications: 'Analysis Ready' (ARD)

Surface reflectance

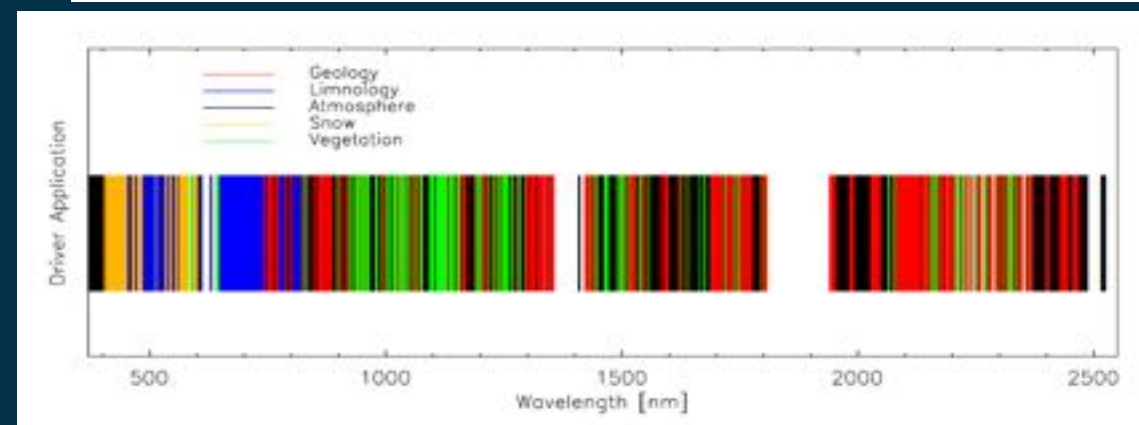
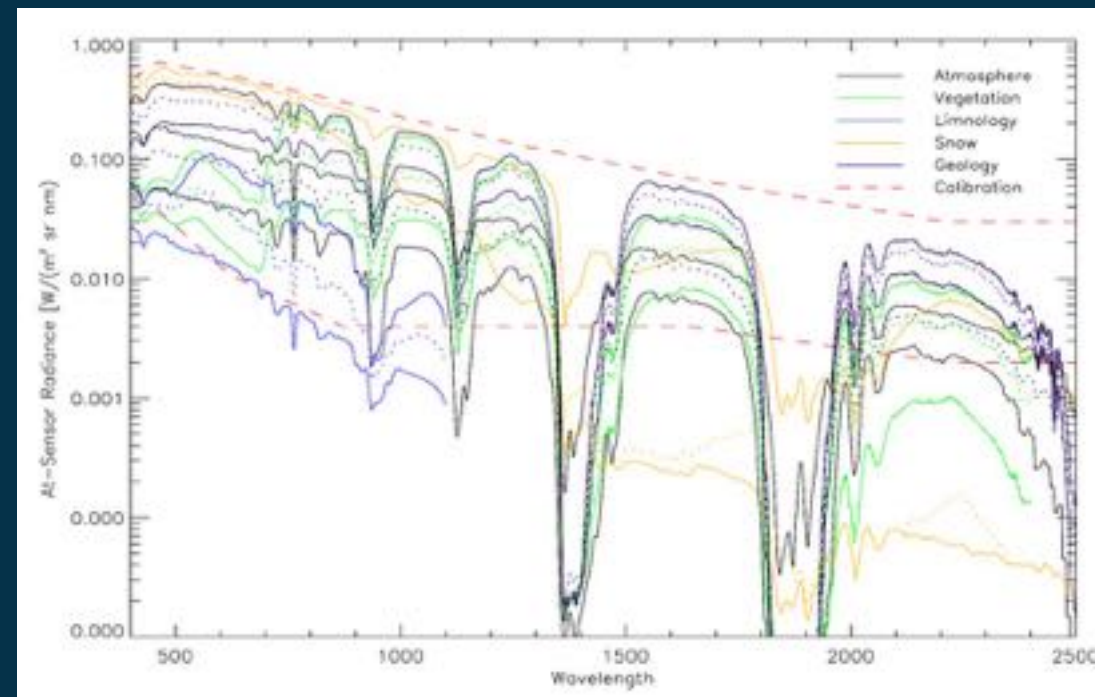
Hyperspectral data can be convolved for many applications enabling an earth system science approach

(not temporally critical applications (61 day repeat – although ~ 20 days for Europe!))

- Directly
- Upgrading other sensors
- Test & Improve retrieval algorithms
- Validation establishing references surface reflectance e.g. Fluxnet

Complementary to EnMAP, PRISMA, CHIME, SBG

- Land-cover change
- Forest
- Surface Albedo
- Agriculture
- Pollution
- Resource prospecting
-



System Drivers: 90 deg precessing 61 day, ~605 km polar orbit for satellite cross-cal & for uniform diurnal sampling for Climate benchmark



Driving requirements in Red

Climate 'Benchmark'

Earth Observation

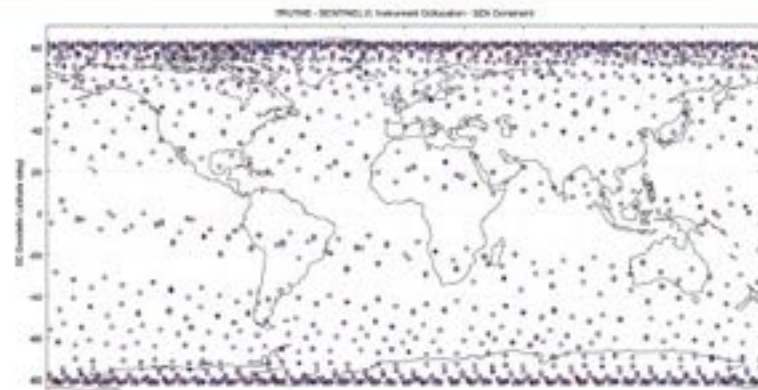


Level 1/2	Mission Requirement						
	Application	Spectral range (nm)	Spectral Resolution (SSI) (nm)	Uncertainty (%) (k=2)		SNR @ sensor SSI & 50 m	GIFOV
				G	T		
Earth spectral radiance (Climate)	350-2200 320-350 2200-2400	<10 - 20 nm	0.3 <1.0	<1.0 <3.0	>~20 @0.3 albedo >15	<300 m (scene ID) 10 - 100 km gridded	
Solar Spectral Irradiance	<320 - 2400	<1 (<400), <5 (<1000) <10 (<2400)	0.3	<1.0	>1000 (time integration)	NA	
Total Solar Irradiance	200-30000	NA	<0.02	<0.05	>10000 (time integration)	NA	
Climate Sensor Cal/Val	320-350 350-800 800-2400	<20 <4 <6-8	0.3	1.0	>15 @0.3 albedo (large footprints)	<300 m--~1000 m(UV)	
Earth sensor (Cal/Val)	<380 ~2400	<4 (< 1000 nm) <8 (>1000 nm)	<1.0	<2.0	>100 @ 0.4 albedo	<50 - 100	
Ocean Colour	<350 -1000	<10 - 20 <4 Cal	<0.5	<2.0	>25 @~0.1 albedo (<400 nm) >50 (400 - 490 nm) >40 (>490 nm)	<300 m Cal/Coastal <~1000 m for Ocean	
Land Surface Imaging	~380-1000 1000-2400	<10-20 <20	<1 <2	<2	>100 @0.3 albedo >60	<50-100 m	
Atmosphere	400 - 2400	<5 - 10	1	2	>100	<100 - 1000	
Lunar Irradiance	350 - 380 380 - 2400	<~20-30 <~10-20	0.3	1	>1000 (time averaged)	NA	

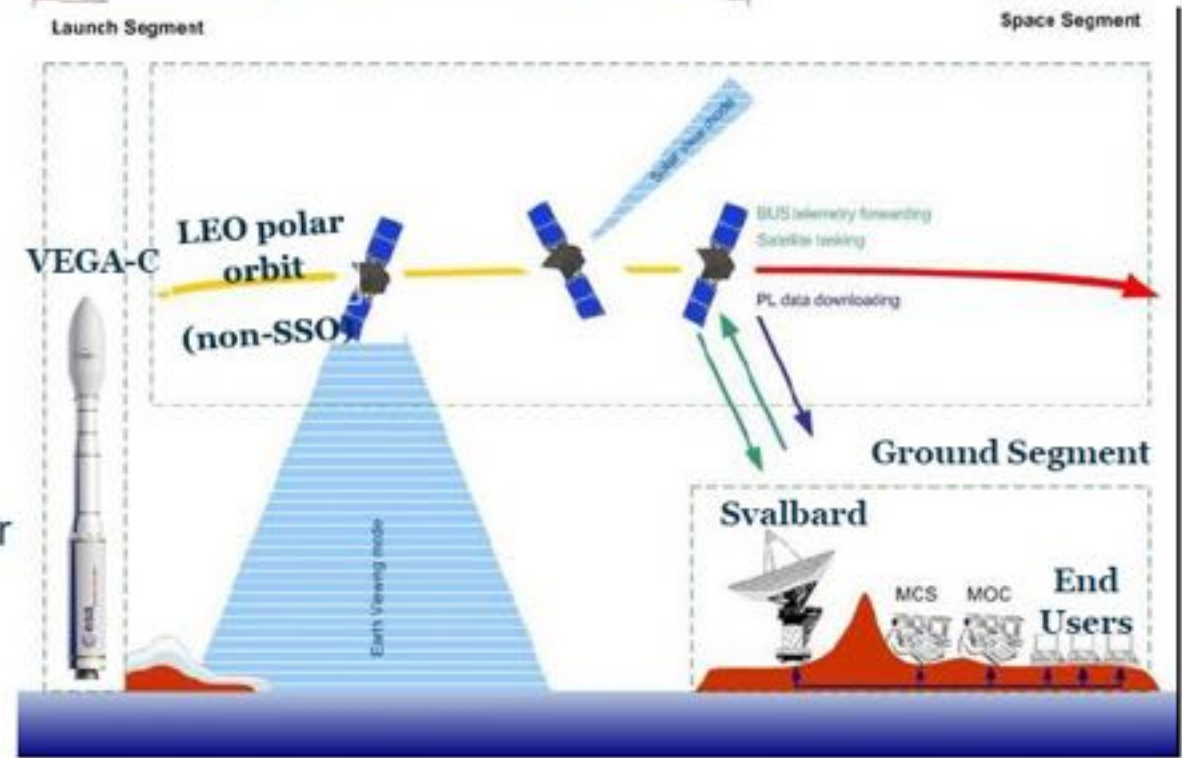


System Architecture

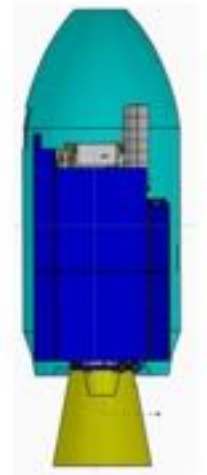
- Lifetime 5 years + 3 extension
- Space Segment:
 - Orbit 614 km, polar (90°) non-SSO
 - 1 satellite – agile, design for non-SSO
 - Novel Payload
- Launch Segment:
 - Vega-C (-E) single launch
 - Option co-passenger explored
- Ground Segment
 - 1 polar station baselined, lossless compression baselined
 - LEOP/early commissioning @ ESOC
 - Routine FOS in UK (Flight control center hand-over before IOCR)
 - PDGS in UK + data access at ESRIN (ESA open data access policy)



Space Segment



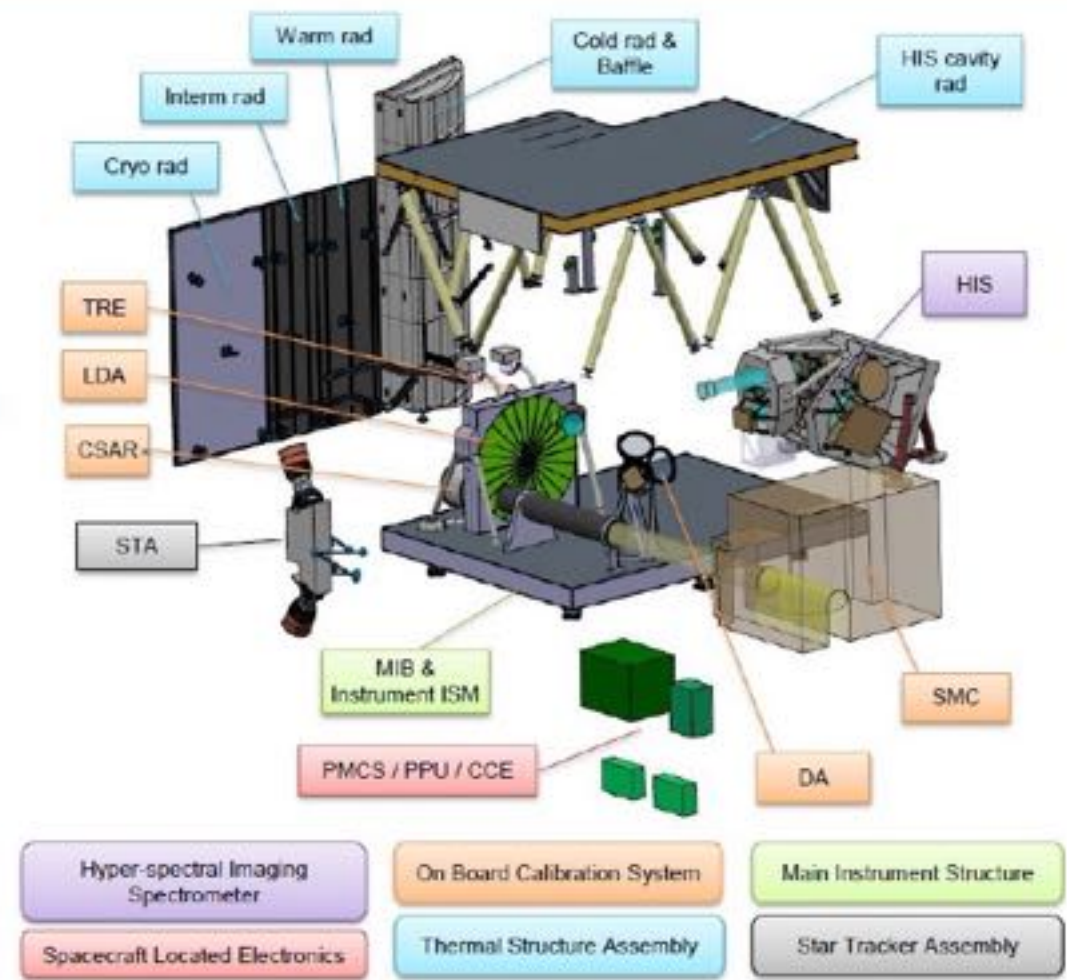
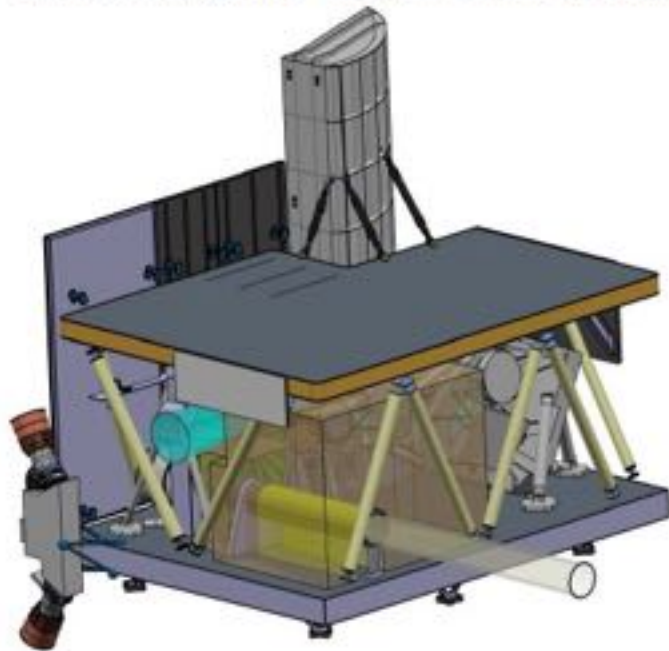
Launch Segment



Payload Overview

Payload, composed of three elements:

- **HIS** (Hyperspectral Imaging Spectrometer) – UV to SWIR (320-2400 nm), single detector, 50 m resolution, 100 km swath. Detector cooled at 150 K passively
- **CSAR** (Cryogenic Solar Absolute Radiometer) – operated at 60 K (cryocooler*), the “primary standard”
- **OBCS** (On-Board Calibration System) – transferring the CSAR solar absolute (SI) measurement to the HIS



*Cryo-cooler Assembly – recurrent from THRISHNA mission at ISRR baseline

Metrology laboratory in-space

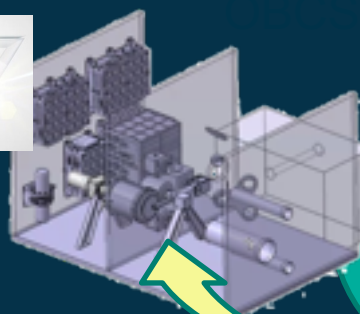
- Measuring energy from the sun, providing the direct traceability to International Standards (CSAR)
 - **Compares heating effect of optical power with electrical power (Volt)**
- ‘Camera’ (Hyperspectral Imaging Spectrometer, HIS) observing the direct incoming and Earth reflected sunlight at high spectral and spatial resolution
- Novel on-board calibration system (OBCS) ensuring traceability to the absolute reference (Cryogenic, 60K, Solar Absolute Radiometer, CSAR) (mimicking terrestrial methods)



Link to SI
= Volt



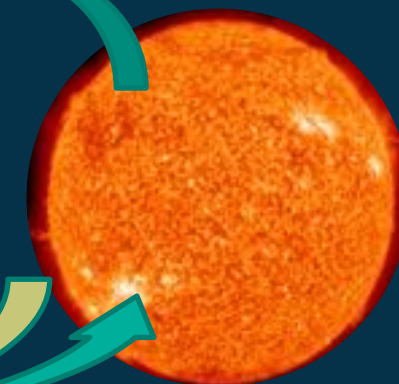
HIS



OBCS



CSAR



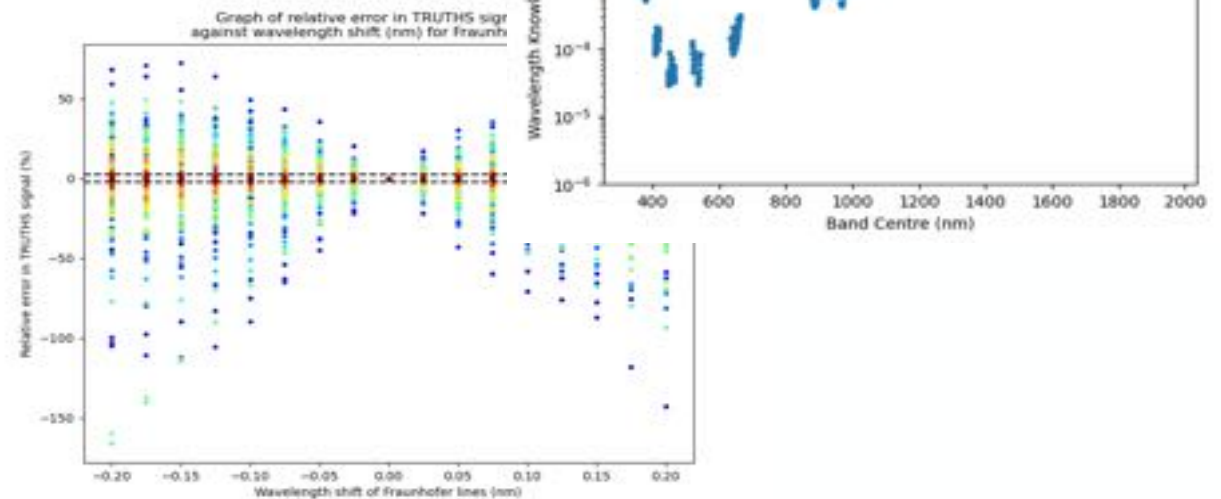
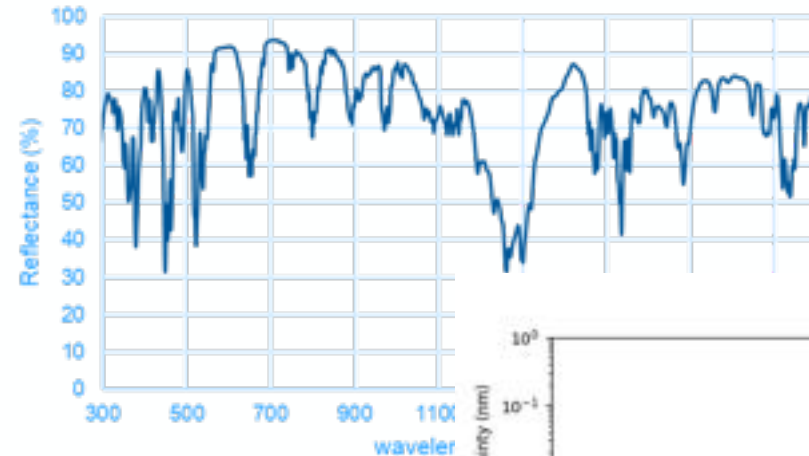
Fraunhofer lines

- 33 identified lines/groups
- 300 nm - 900 nm
- Use Solar spectral irradiance observation – few seconds
- <math><0.025\text{ nm}</math> accuracy est.

Onboard Rare-earth oxide doped filter reference

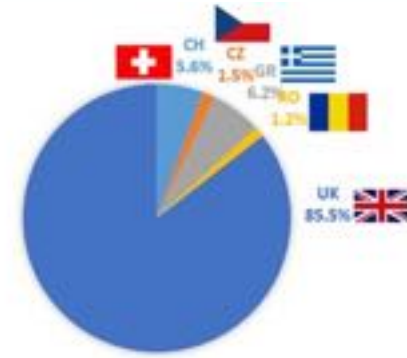
- Features over 300 – 1950 nm
- Solar illuminated convolved with Fraunhofer lines, but feature width allows reliable determination.
- <math><0.1\text{ nm}</math> accuracy est.

Overlap region adds confidence.

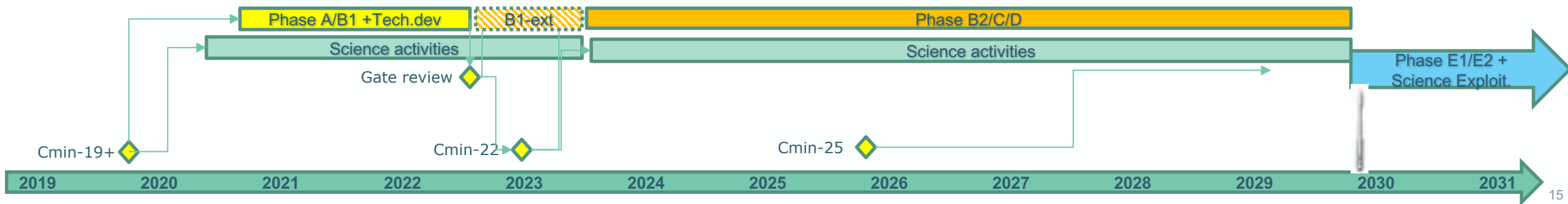


TRUTHS Timeline

- TRUTHS was proposed by UKSA in May 2019 as a new Earth Watch (EW) Element.
- TRUTHS Phase A/B1 has been fully subscribed at Space19+ by 5 Participating Countries: UK (85.5%), GR (6.2%), CH (5.6%), CZ (1.5%), RO (1.2%)
- Industrial Phase A/B1 system studies and technology predevelopments initiated in Oct-20.
 - Phase-A kicked-off in Oct-20 and completed at end-July 2021
 - Phase B1 completed in Q2-2022 _ extended for technology maturing and to bridge to B2 phase to start in 2023. (TRL 5/6 by Phase B2 and SRL 5)
- Mission Advisory Group (MAG): Science/Engineering/User expertise primarily from Europe (not limited to funding nations) inc NASA CLARREO Pathfinder
- Programmatic “Gate Review”: go decision, taken in July-22, to submit program to CM-22
- Phase B2/C/D/E to be funded at CM-22/-25 -> Program plan being currently prepared



TRUTHS A/B1 SUBSCRIPTION - @SPACE19+



International (CEOS?) climate & calibration Observatory



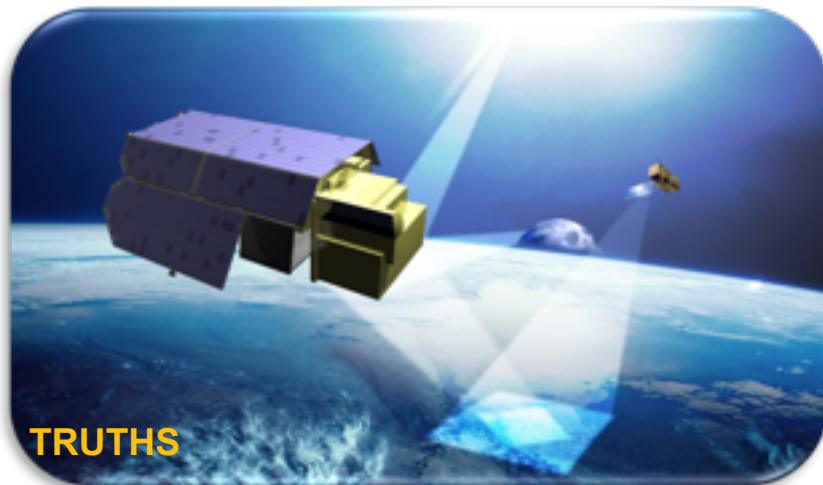
TRUTHS ~ 2029/30 will help initiate a sustainable long-term international climate & calibration observatory

A direct response to international requests

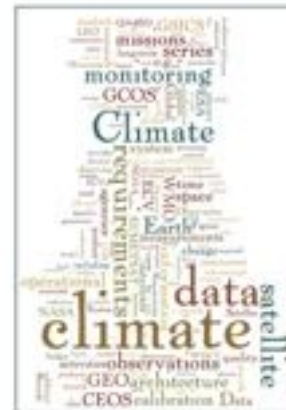
NASA CLARREO-Pathfinder 'sister mission' which will be launched to the ISS in 2023/24.

- Hope for overlap!
- Also potential Chinese Libra

TRUTHS & CPF SITSats will provide unique and critical information for understanding and monitoring Climate and environmental change from space and support climate action – A resource for ALL nations



Strategy Towards an Architecture for Climate Monitoring from Space



SI-Traceable Space-based Climate Observing System: a CEOS and GSICS Workshop National Physical Laboratory, London, UK, 9-11 Sept. 2019

